Mitigation of ground motion in ATF2

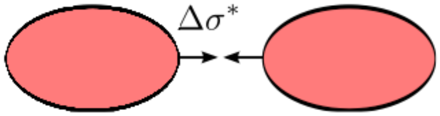
J. Pfingstner, K. Artoos, C. Charrondiere, St. Janssens, M. Patecki*, Y. Renier, D. Schulte, R. Tomas (*CERN*),
A. Jeremie (*LAPP-IN2P3-CNRS*),
K. Kubo, S. Kuroda, T. Naito, T. Okugi, T. Tauchi, N. Terunuma (*KEK*)
**Also at Warsaw University of Technology, Department of Physics*

Outline

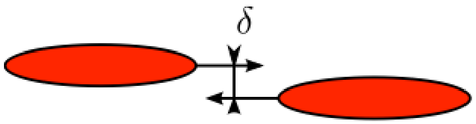
- Motivation
- Mitigation concept modelling
- Performance predictions
- Simulation studies for the ATF2 GM experiment
- Experimental setup in ATF2
- Ground motion measurements in ATF2
- Correlation results
- Orbit jitter reduction
- Conclusions

Motivation

- The future particle accelerators are going to be more and more sensitive to ground motion (GM) effects.
- GM causes the accelerator magnets misalignments which can result in:
  - the beam size growth:



  - the beam-beam offset at the IP:

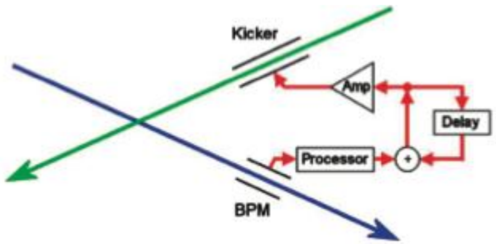


Motivation (II)

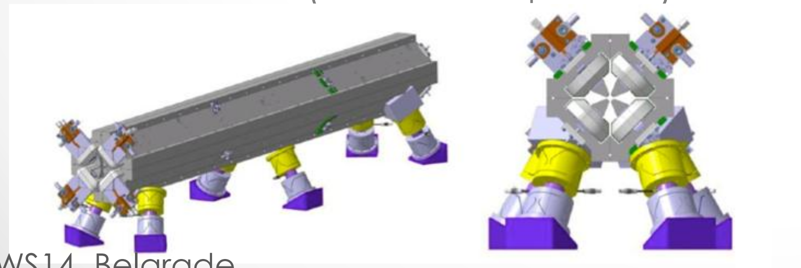
- The GM effects can be suppressed by the orbit feedback systems, which are efficient for low frequencies (factor $f_R/20$ by the rule of thumb).
- For ILC f_R= 5 Hz
- For CLIC f_R= 50 Hz
- For ATF2 f_R= 3.12 Hz
- **The orbit feedbacks are not sufficient to suppress all relevant ground motion effects.**

Motivation (III)

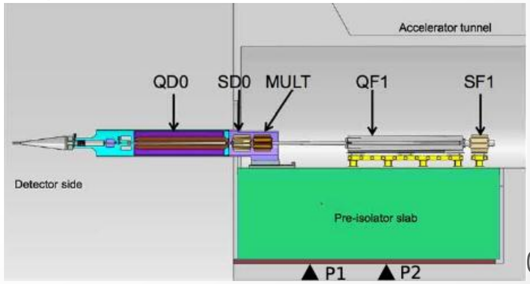
- For the frequencies (higher than about $f_R/20$) not corrected by the orbit feedback system, there are other correction methods:
  - Intra-train feedback systems - J. Resta-Lopez, P. Burrows, and G. Christian, Journalof Instrumentation 5, 09007 (2010).



    - More efficient for ILC than CLIC (short bunch spacing (0.5 ns))
    - May cause the luminosity loss
  - Mechanical (active and passive) stabilization systems



K. Artoos et al.



A. Gaddi et al.

Motivation (IV)

- For the frequencies (higher than about $f_R/20$) not corrected by the orbit feedback system, there are other correction methods:

Feed-forward system based on Ground Motion

- Cheaper than mechanical systems,
- Easy to integrate into accelerator modules,
- Possibility to apply corrections that are distributed over many correctors (global scheme),
- High demands on the speed of the system,
- High demands on the system model accuracy.

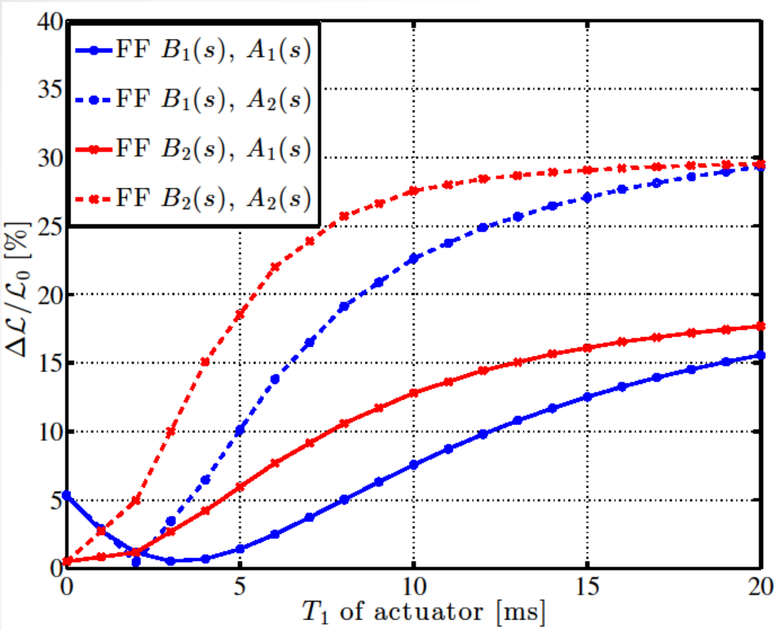
GM feed-forward system

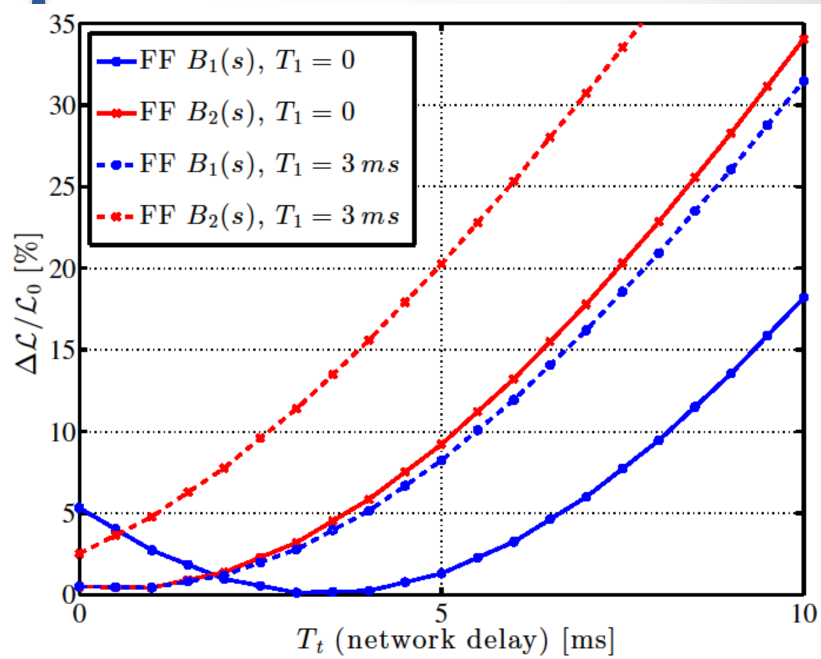
- The vibrations are measured by the sensors in order to determine in real-time the quadrupoles position change $\boldsymbol{x}(t)$.
- The beam orbit change $\boldsymbol{b}(t)$ is predicted with the use of orbit response matrix $\boldsymbol{R}_q$

$$\boldsymbol{b}(t) = \boldsymbol{R}_q \boldsymbol{x}(t).$$

- The actuations $\mathbf{c}(t)$ of corrector magnets that compensate $\boldsymbol{b}(t)$.

Performance predictions





- $B_1(s)$ is a transfer function of the seismometer CMG-6T from Guralp with cut-off frequencies of 0.03Hz and 100Hz
- $B_2(s)$ is a transfer function of the geophone from CLIC stabilization group with cut-off frequencies of 0.9Hz and 1kHz
- Two types of actuators:

$$A_1 = \frac{1}{1 + T_1 s},\ A_2 = \frac{1}{(1 + T_1 s)^2}$$

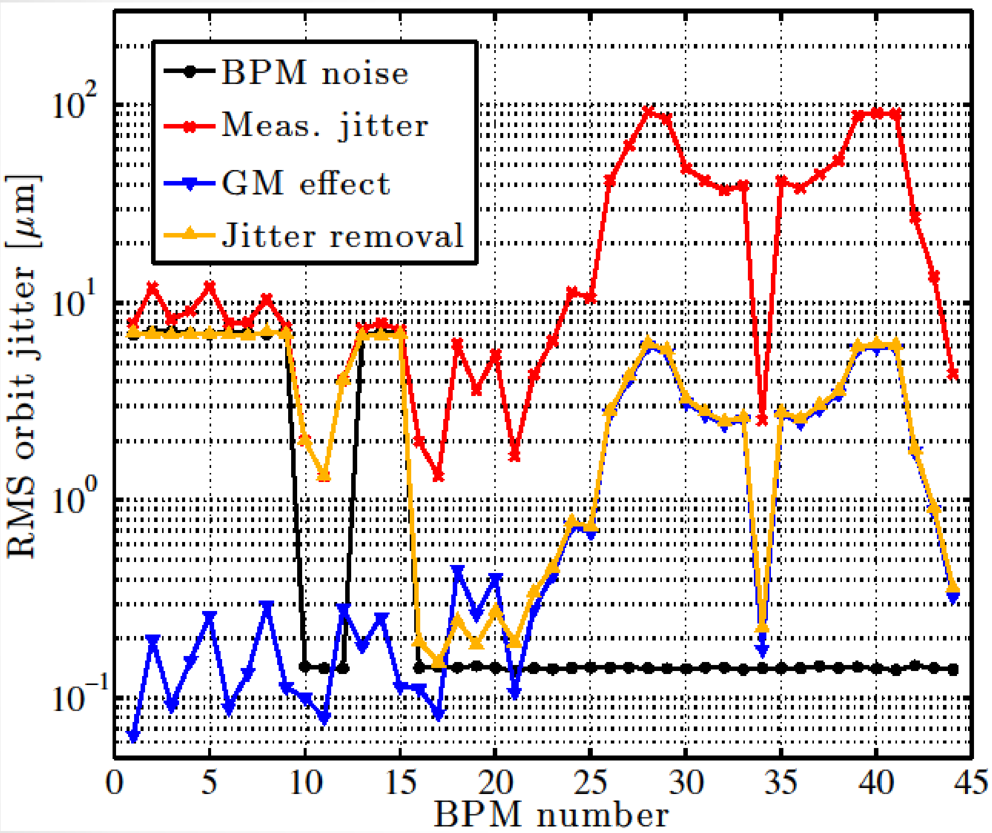
T_1 is a time constant corresponding to the time to reach 63.2% of the amplitude of an applied step function.

- **Seismometer CMG-6T from Guralp is chosen for the experiment**

Simulation studies for the ATF2 GM experiment

- The simulations were conducted in order to evaluate the expectable performance of the feed-forward mitigation method.
- PLACET and ATF2 Flight Simulator were used showing good agreement.
- GM generator integrated into PLACET.
- Two setups with 14 and 30 sensors located in the optimized positions.
- Quadrupole magnets field errors of 0.01%,
- BPM scaling errors of 1%,
- Stripline BPMs resolution set to 5μm, and cavity BPMs to 0.1 μm
- Initial orbit jitter with an amplitude of 10% (horizontal) and 25% (vertical),
- Sextupole magnets turned off.

Simulated orbit jitter



- Low resolution of BPMs at the beginning of the beam line,
- GM effect very small comparing to the full orbit jitter,
- Incoming orbit jitter removed by the de-correlation technique:

Incoming orbit jitter can be measured by high-resolution BPMs (10,11,12) and then removed from the downstream BPMs:

$$\Delta B_i^{(r)} = \Delta B_i - K_{up} \Delta B_i$$

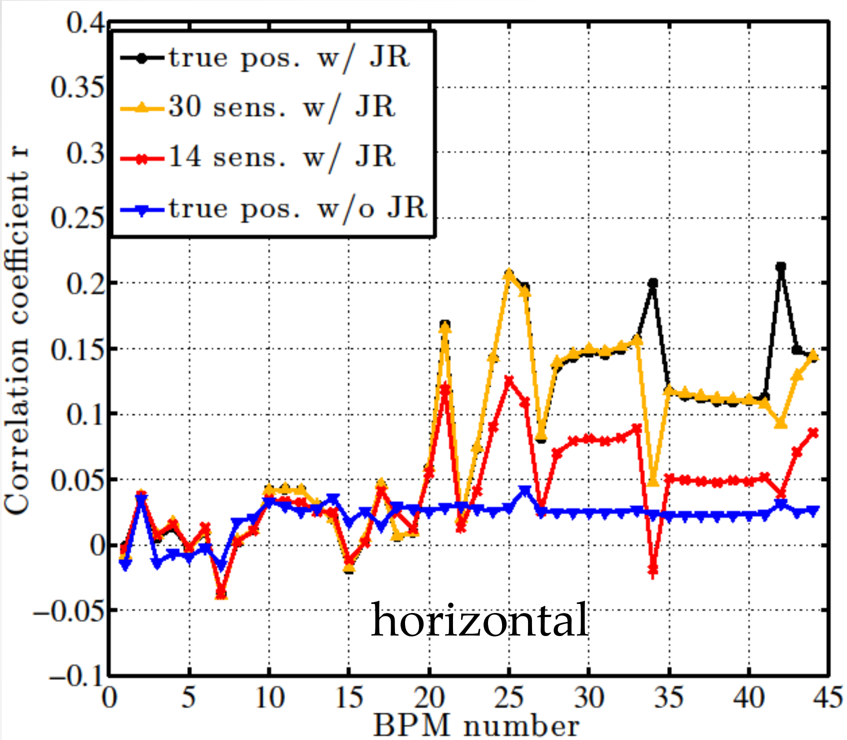
$$K_{up} = \Delta B_{up} \Delta B_{up}^{\dagger}$$

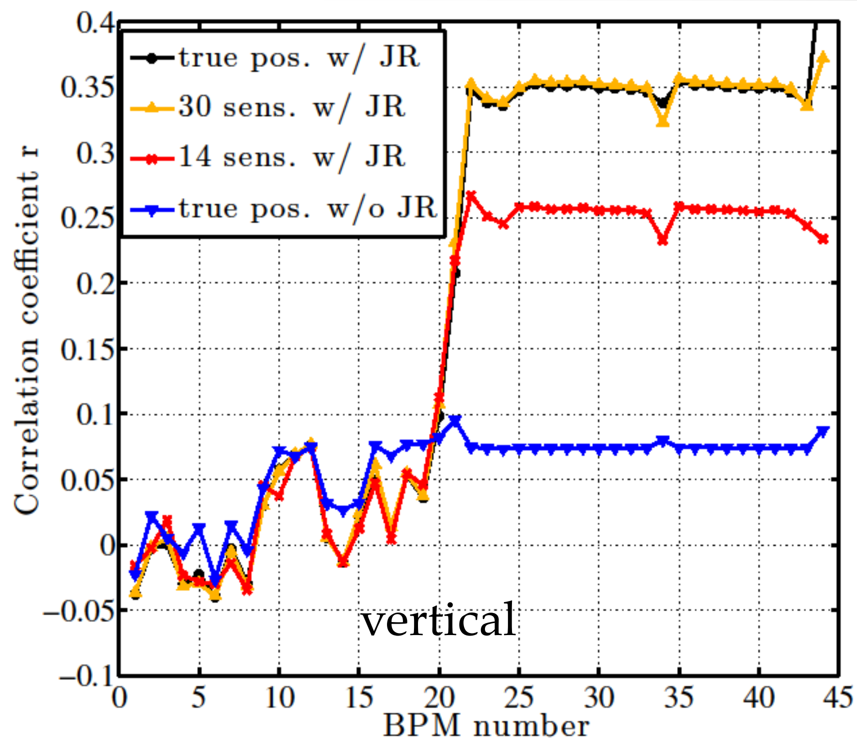
$$\Delta B_{up} = [B_{10}, B_{11}, B_{12}]$$

$\Delta B_i^{(r)}$ are the decorrelated data with the incoming jitter removed, and † stands for the pseudo-inverse of a matrix.

- **The ground orbit effect can be measured after the jitter removal.**

Correlation coefficient

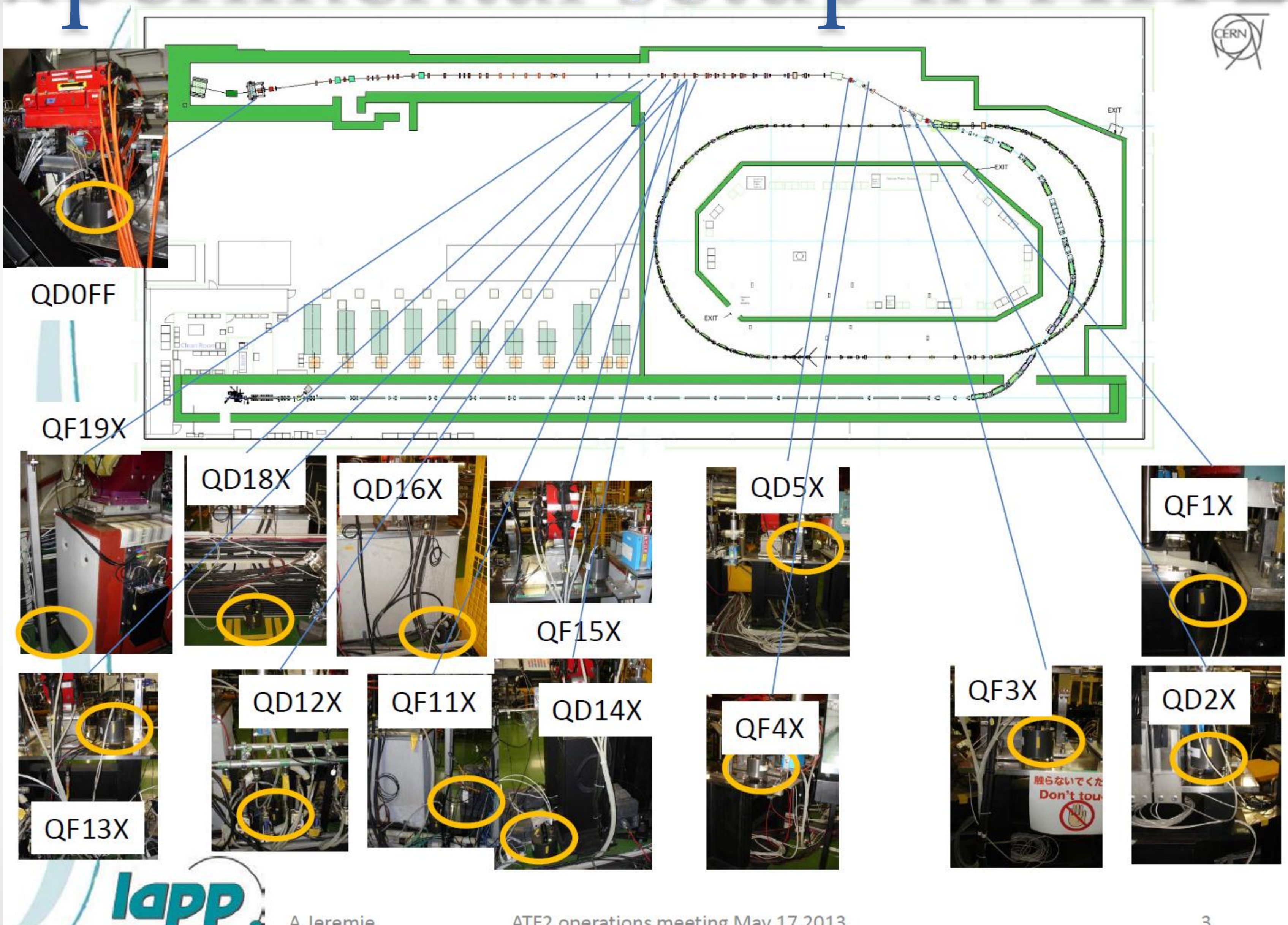




- $\Delta b_k = b_k - b_{k-1}$ - BPM data (high pass filtering)
- $\Delta\widehat{x_k}$ - estimated position changes of quadrupoles
- $\Delta\widehat{b_k} = R_q \Delta\widehat{x_k}$ - predicted beam orbit at BPM locations

- $r_i = \frac{cov(\Delta B_i, \Delta\widehat{B_i})}{\sigma(\Delta B_i)\sigma(\Delta\widehat{B_i})}$ - **correlation coefficient**
- $\Delta B, \Delta\hat{B}$ - matrices for measured and predicted beam orbit for all BPMs and time steps
- $\sigma(b_i)$ – standard deviation of any vector b_i
- $cov(b_i, b_j) = \frac{1}{1-N}\sum_{k=1}^{N}(b_i(k) - \overline{b_i})\,(b_j(k) - \overline{b_j}))$

Experimental setup in ATF2



Experimental setup in ATF2

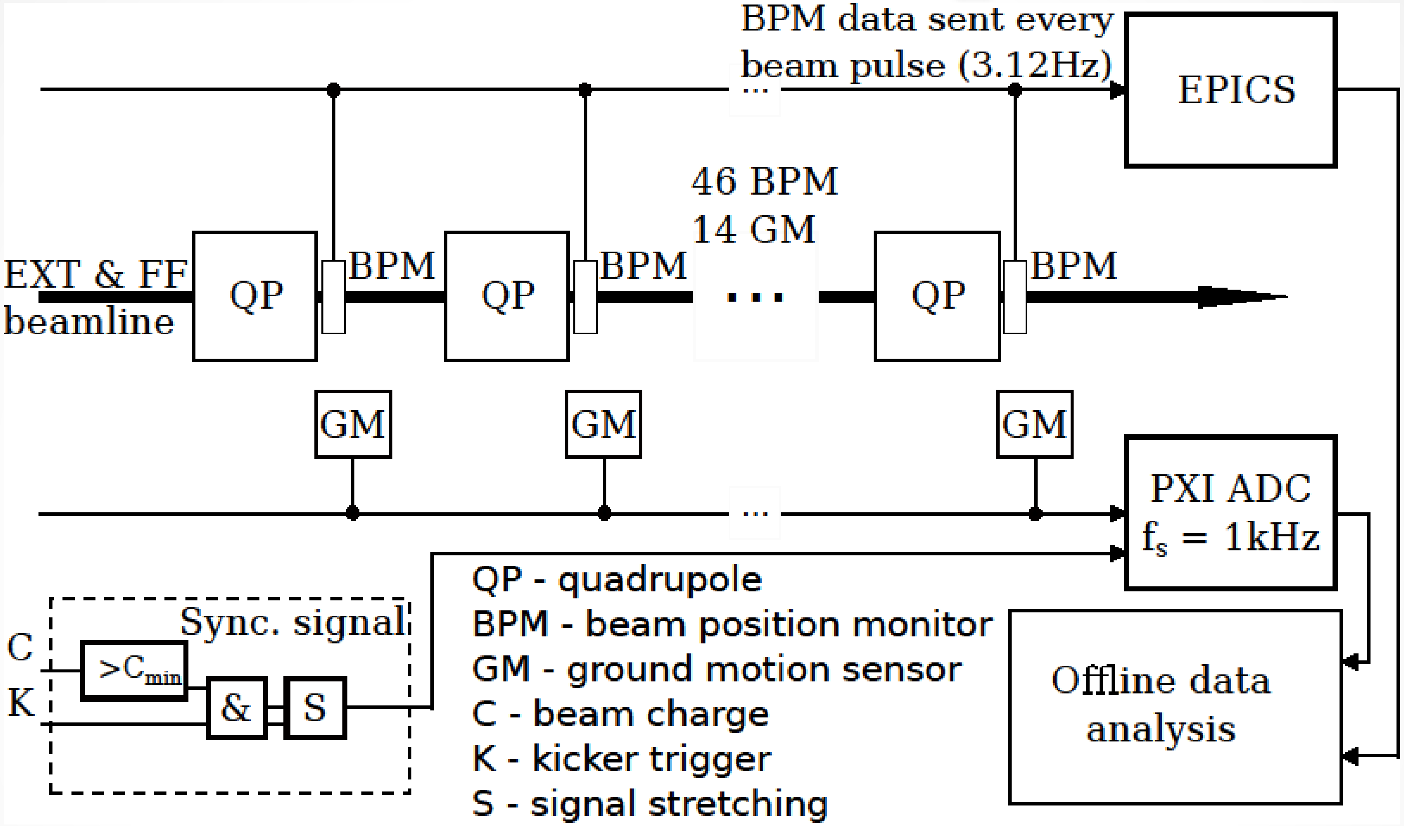
Velocity output bandwidth	1 s – 100 Hz (Model CMG-6T-1), 10 s – 100 Hz (Standard) or 30 s – 100 Hz
Velocity output sensitivity	2 × 1200 V/m/s, (Standard) 2 × 2000 V/m/s or 2 × 1000 V/m/s
Peak output	±10 V (20 V peak-to-peak)
ptional high gain sensitivity	2 × 10000 V/m/s (adjustable)
Lowest spurious resonance	450 Hz
Linearity	> 90 dB
Cross-axis rejection	> 65 dB
Electronics noise level	–172 dB (rel. 1m2s-4Hz-1)
Operating temperature	–40 to +75 °C
Temperature sensitivity	< 0.6 V per 10 °C
Mass recentring range	±3 ° from horizontal
Materials	Hard anodised aluminium case Gold plated contacts O-ring seals throughout
Case diameter	154 mm
Case height (with handle)	207 mm
Weight	2.49 kg
Power supply	10 – 36 V DC
Optional low power sensor	5 V DC supply (output ±4.5 V)
Current at 12 V DC	38 mA

Name	Location	Distance from S1
S1	on QF1X	0.0m
S1	on QD2X	2.0m
S3	beside QF3X	3.0m
S4	beside QF4X	7.3m
S5	beside QD5X	8.2m
S6	beside QF11X	22.0m
S7	beside QD12X	23.4m
S8	beside QF13X	24.5m
S9	beside QD14X	25.5m
S10	beside QF15X	26.5m
S11	beside QD16X	27.6m
S12	beside QD18X	31.4m
S13	beside QD19X	32.4m
S14	beside QD0FF	80.8m

GURALP SYSTEMS

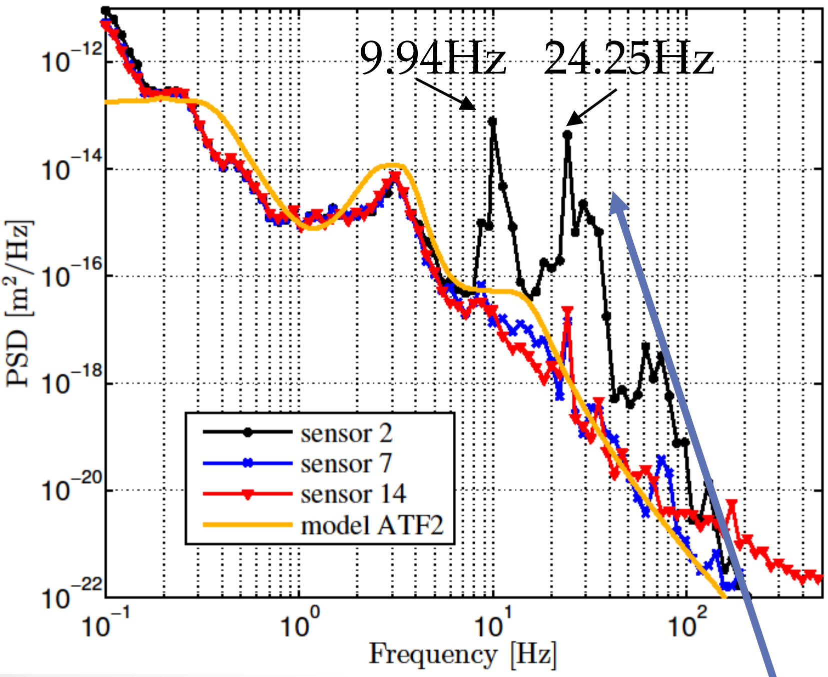
CMG-6T

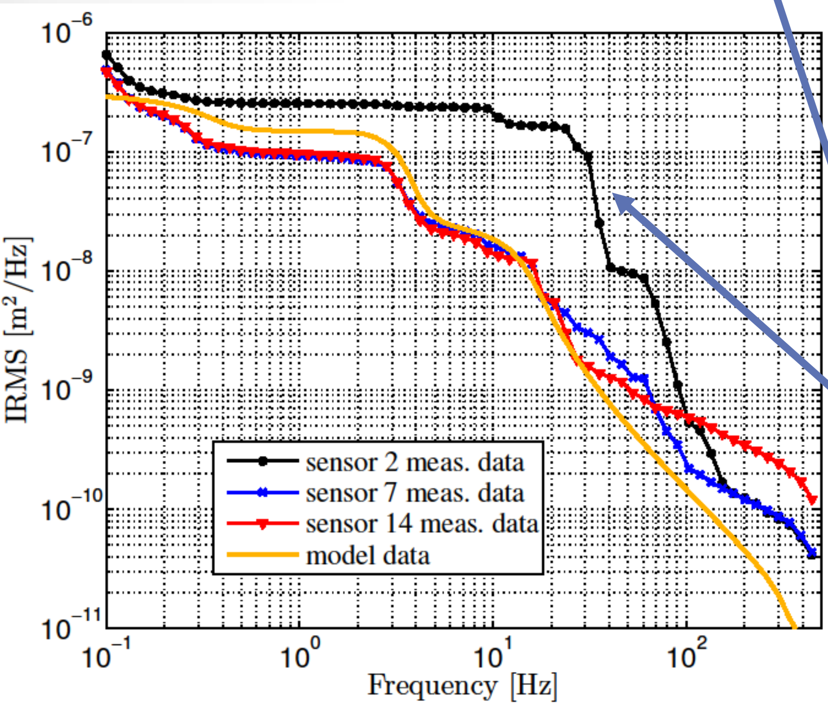
Experimental setup in ATF2

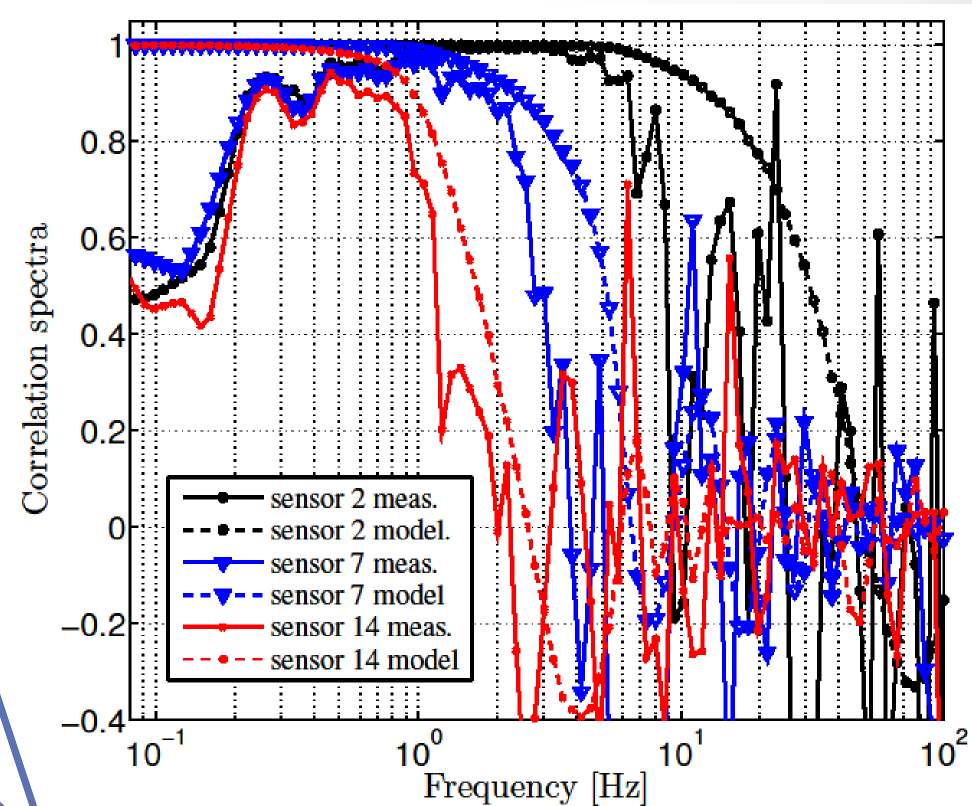


- 46 BPMs for the beam orbit measurements. Data sent to EPICS every beam arrival (3.12 Hz),
- 14 GM sensors connected to NI PXI 8109 RT via low noise cables and digitized by a card NI 6289 with sampling frequency of 1024 Hz,
- Synchronization signal recorded by PXI enabling to select the GM data corresponding to BPM data.

Ground motion measurements

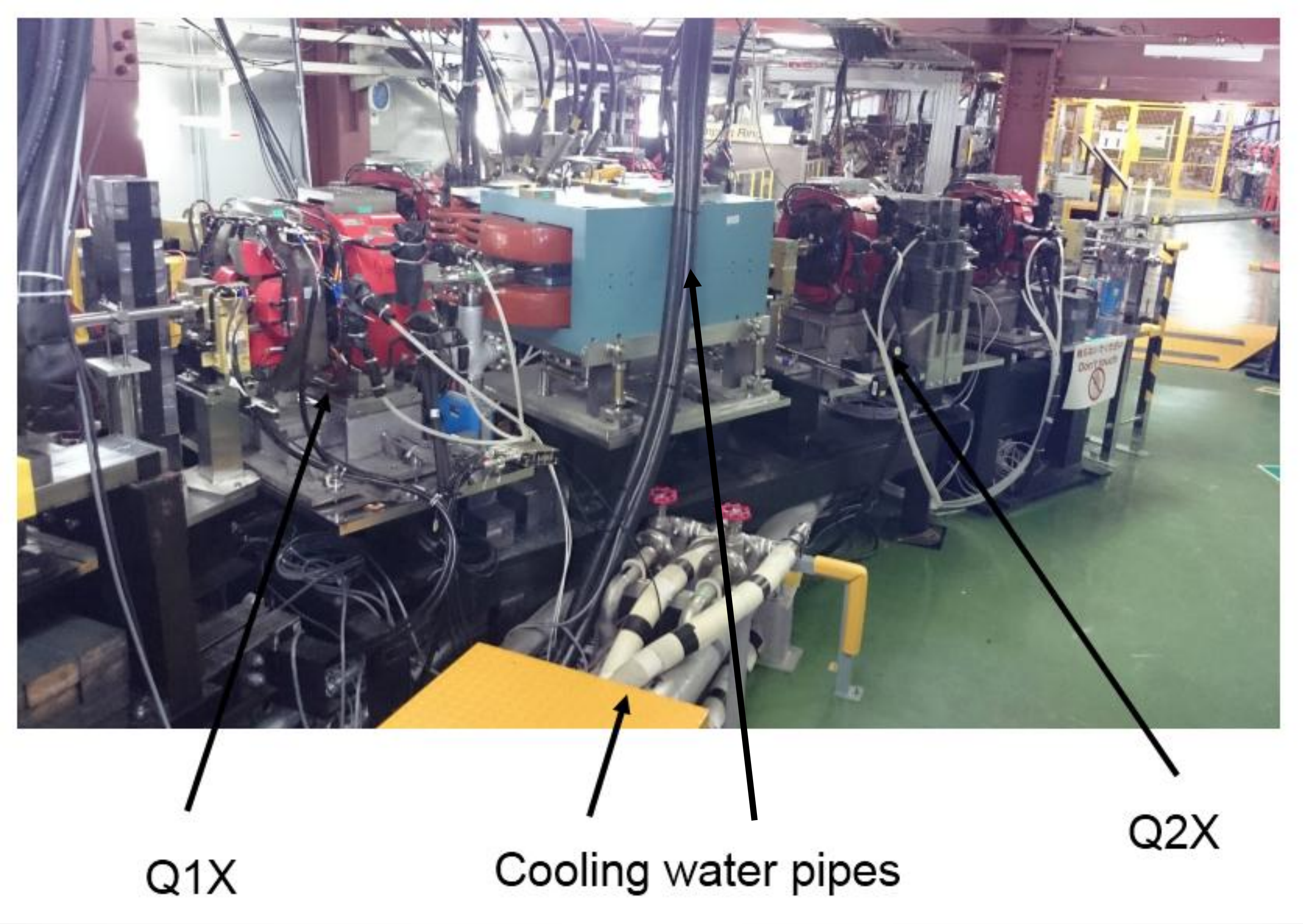




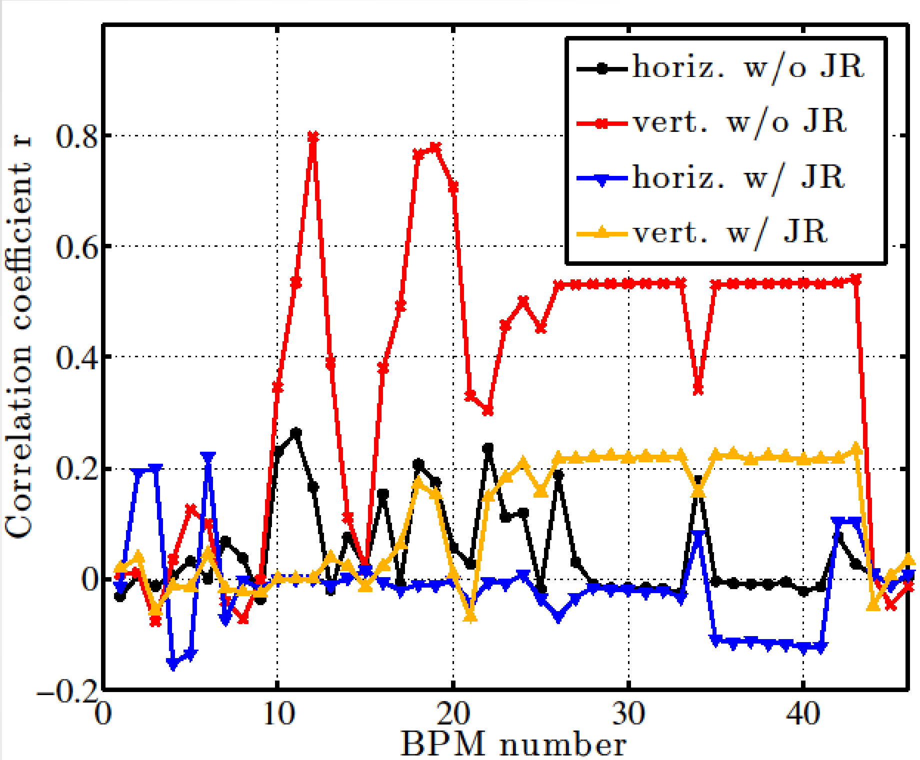


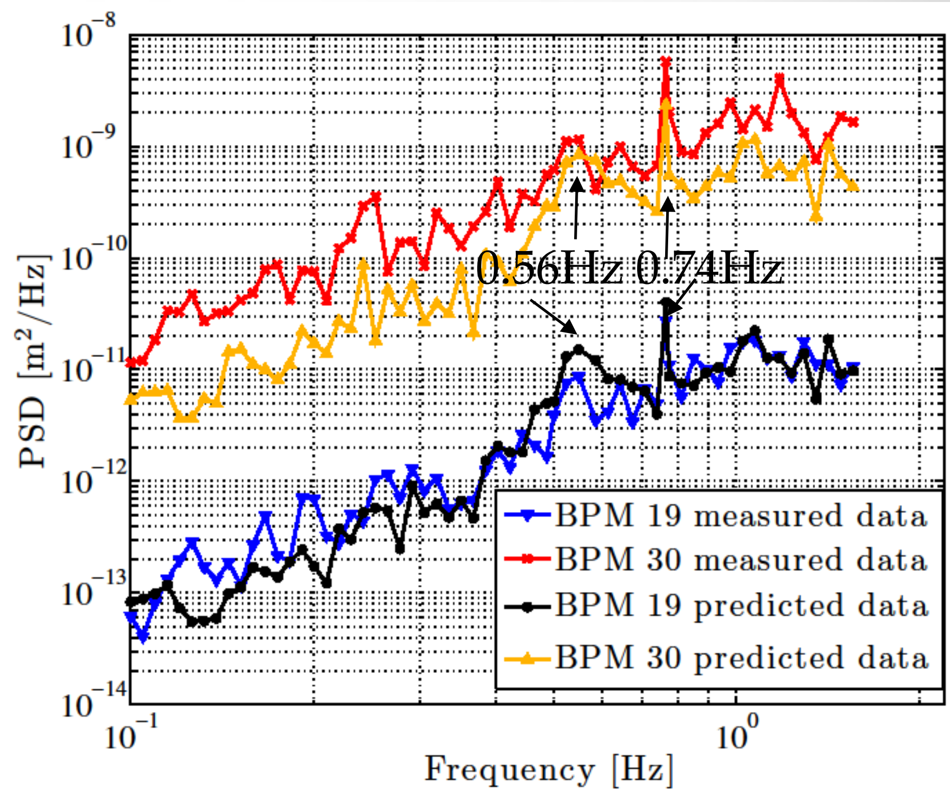
High, unexpected vibration around sensor 2

Sensor 2 region



Correlation coefficient

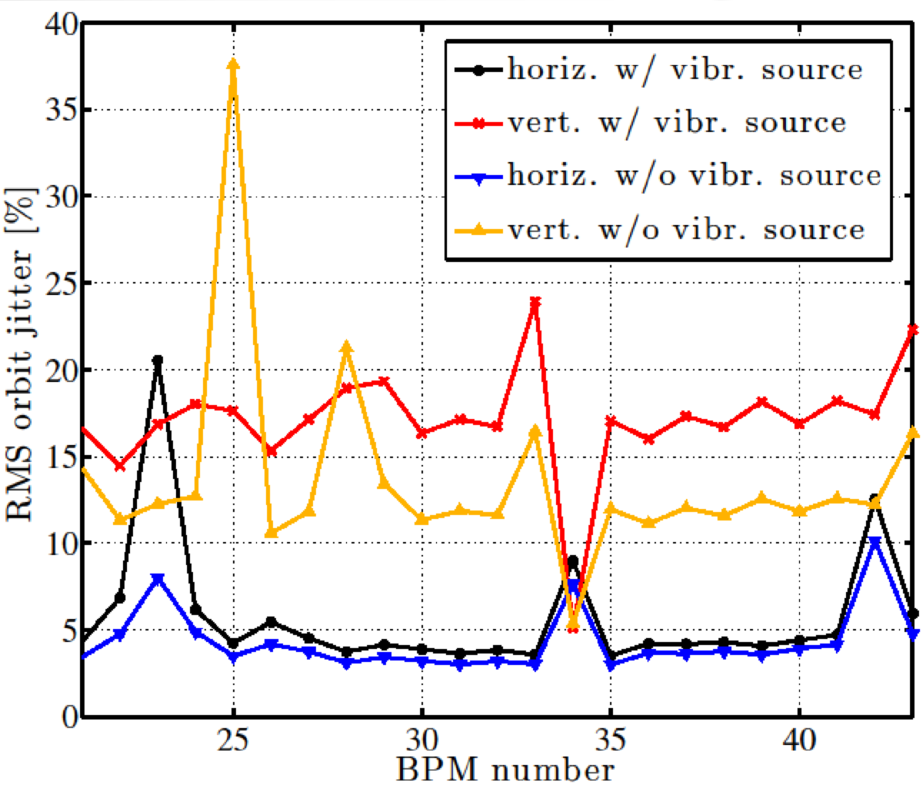


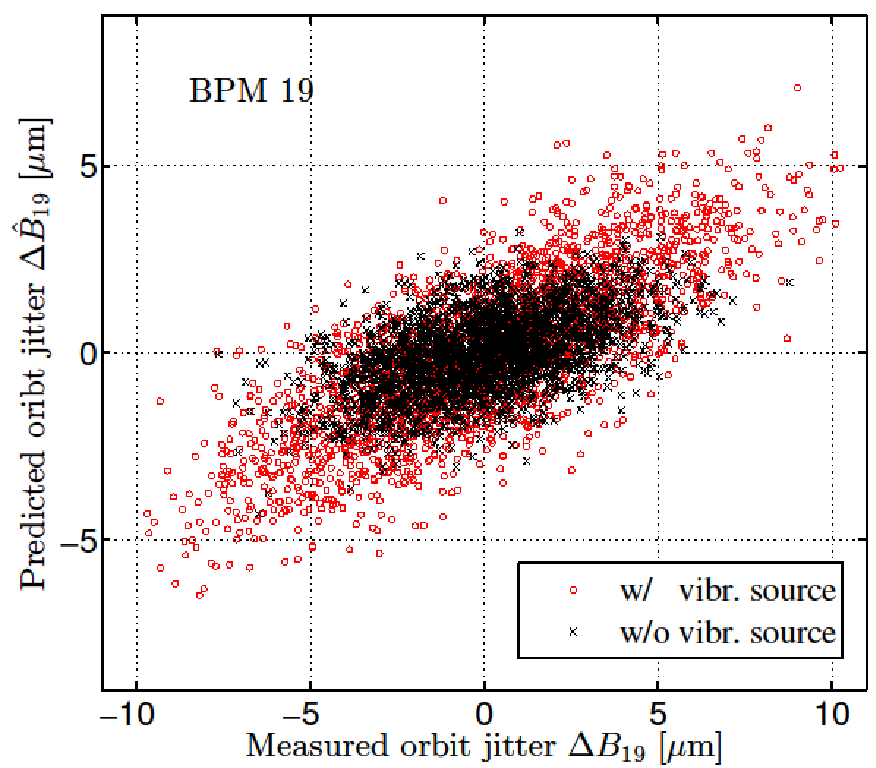


- Very high correlation between actual and predicted beam orbit.
- Still some correlation after applying jitter reduction (JR).
- **GM sensors can be used for the orbit distortions predictions.**

- Correlation also visible in frequency space of BPM data.
- Two peaks at 0.56Hz and 0.74Hz correspond to peaks in PSD of GM sensor 2 (9.94Hz and 24.25Hz).

Orbit jitter removal





- After removing the vibration source around sensor 2, the jitter level was decreased by a factor of 1.4 in amplitude which corresponds to halving the orbit jitter power.
- The correlation between measured and predicted orbit jitter for BPM 19. The correlation visibly decreased after removing the vibration source. Mind the scaling factor probably coming from imperfect response matrix R_q used for predicting the beam orbit aberrations.

Summary

- Novel mitigation method based on GM-feed-forward system is studied in order to counteract the ground motion impact on the beam.
- It was experimentally proved that the GM sensors can be used for beam orbit distortions predictions.
- One vibration source in the ATF2 beam line was found and eliminated which shows the potential of the system to discover the installation issues and model mismatches.
- The real-time GM-feed-forward system needs the fast electronics in order to collect the GM data, calculate the correction and apply it to actuators.

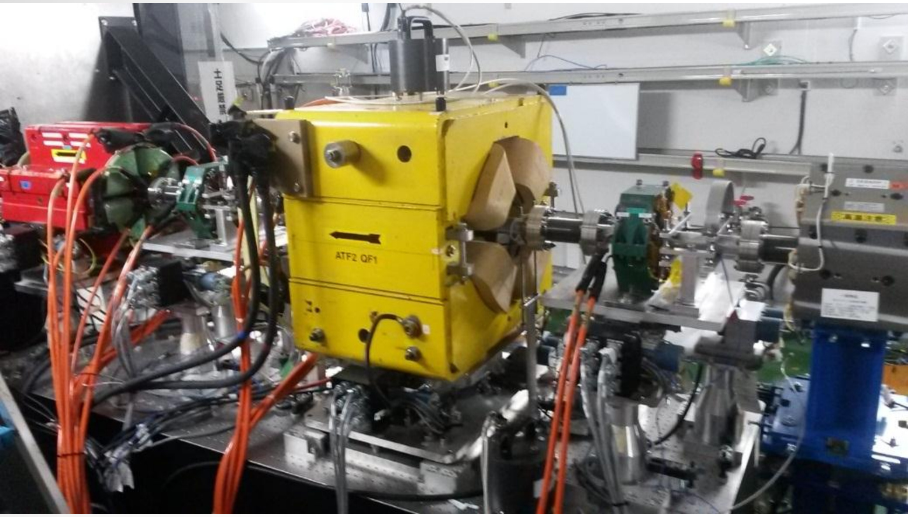
GM measurements 29.09 – 12.10

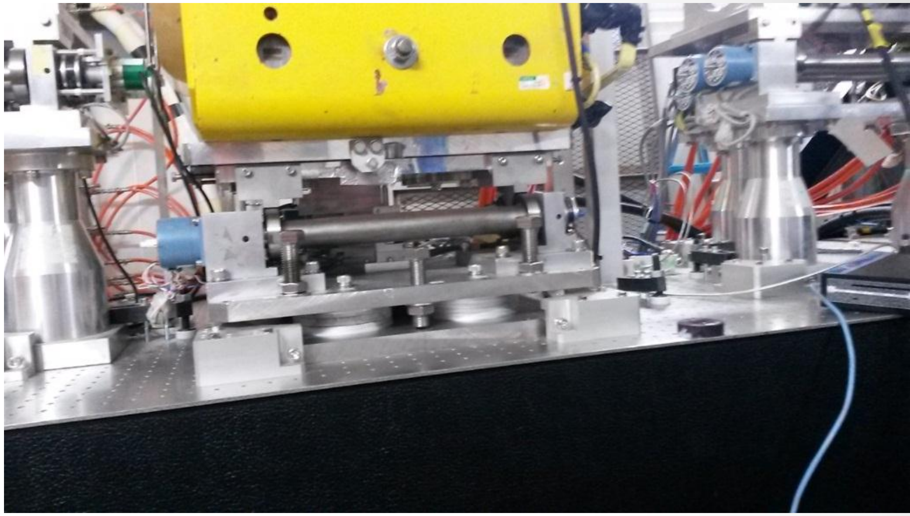
A. Jeremie (*LAPP-IN2P3-CNRS*),
T. Okugi (KEK) and M. Patecki (CERN, WUT)

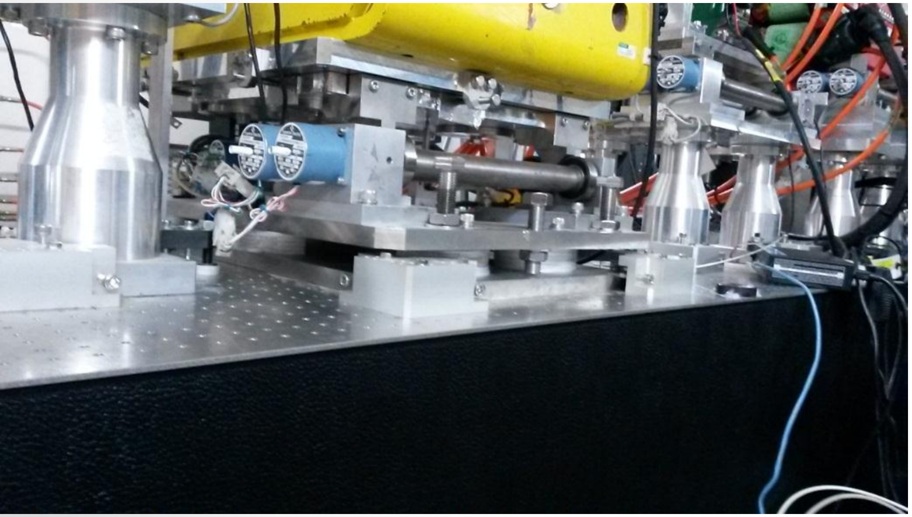
Plan of measurements

- Vibrations of QF1FF relative to the table and ground,
- Investigation of support stability,
- Vibrations around QD10ABFF (near to cooling water pipes),
- Coherence length from FD.

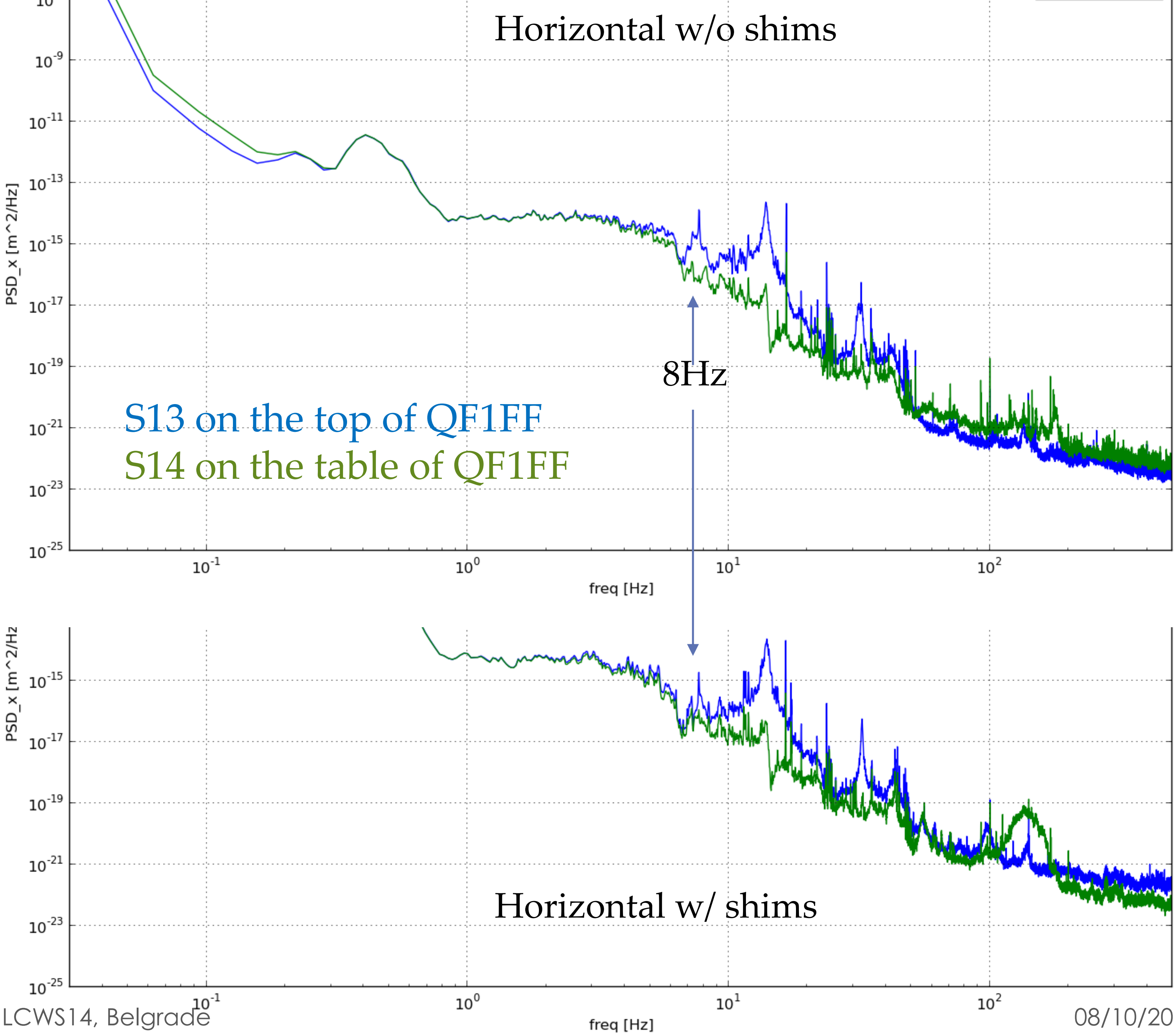
QF1 support



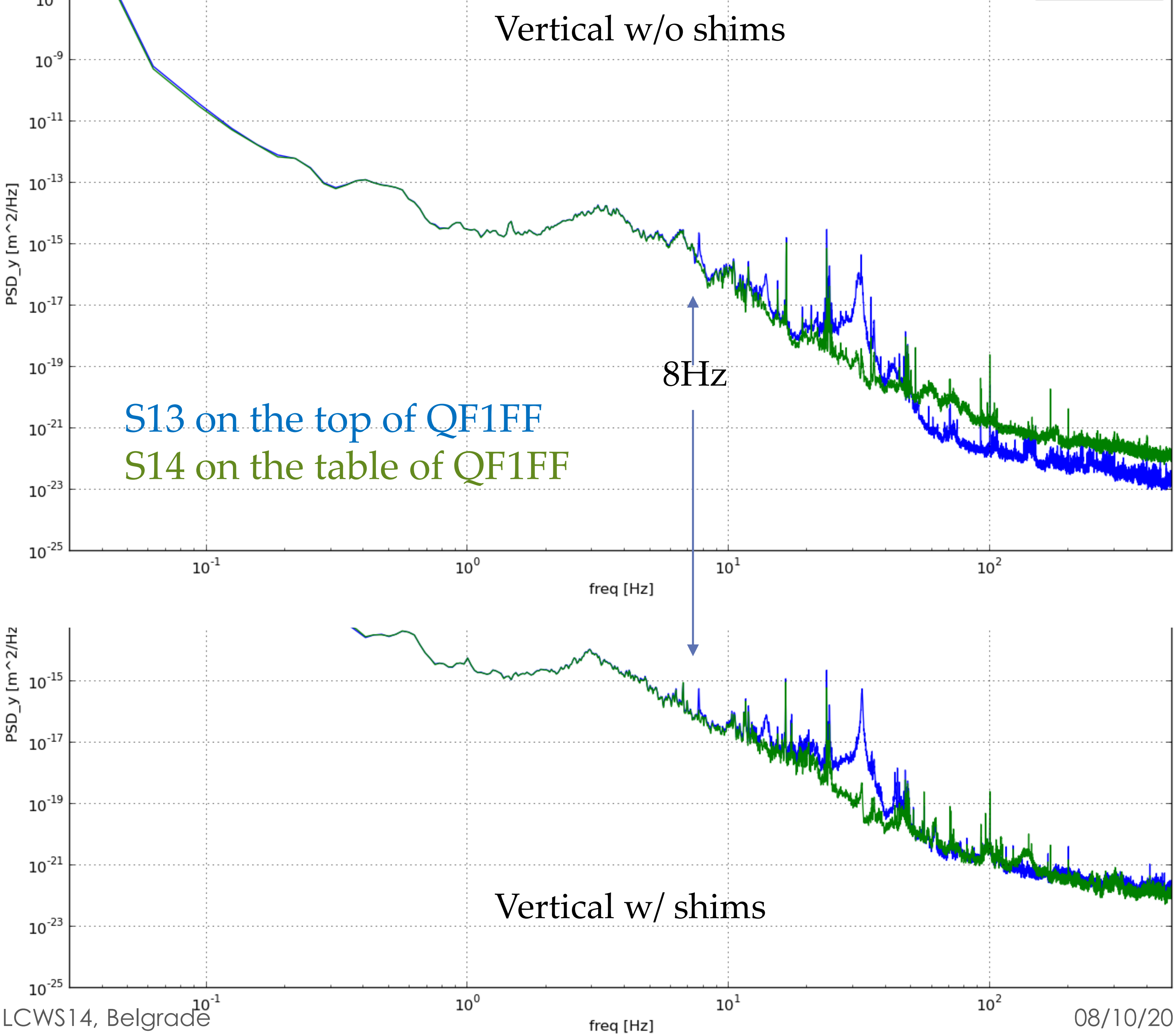




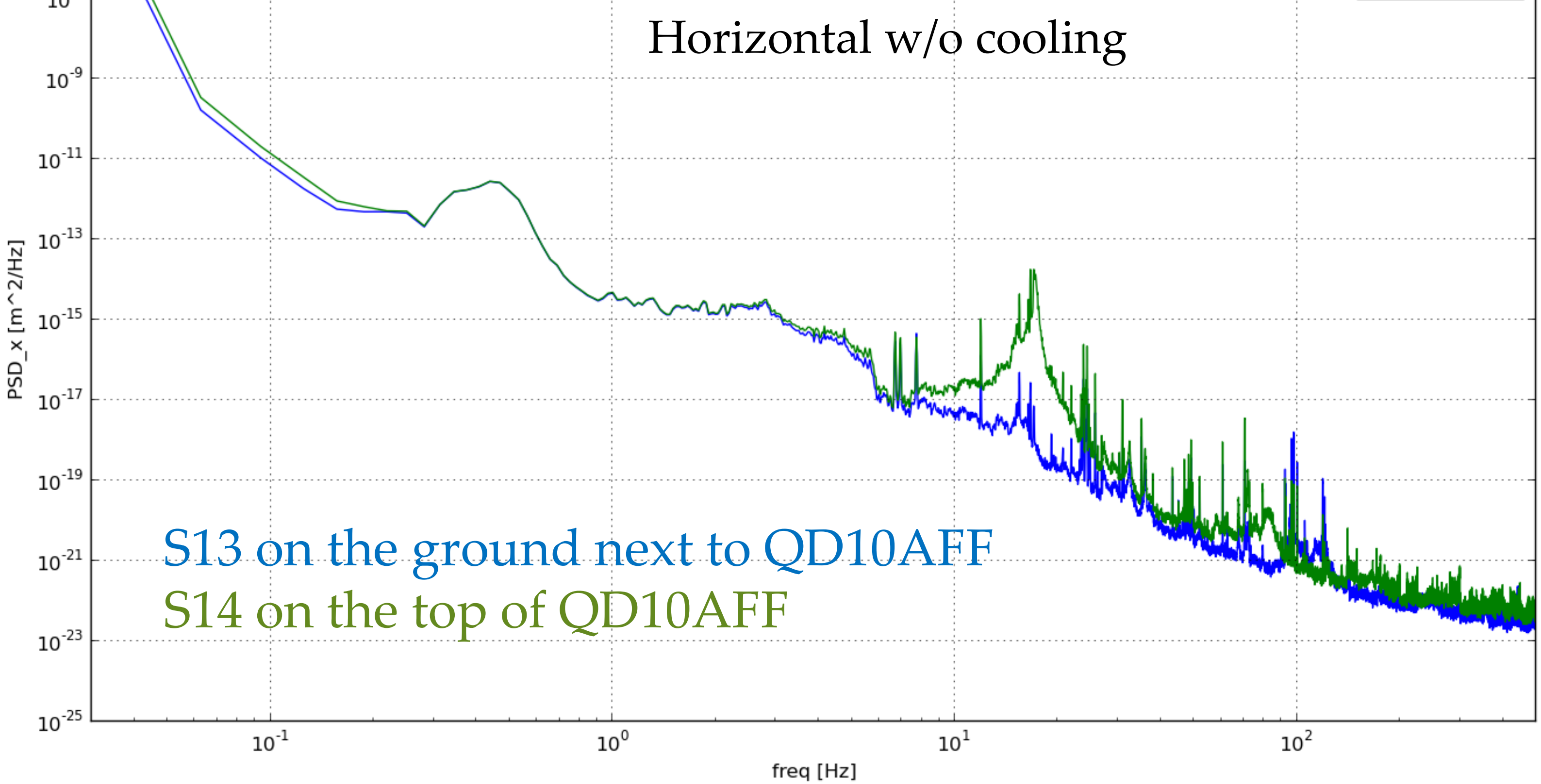
- A new QF1 is big and heavy – 2-3 times heavier than before
- Support of QF1 could be not solid enough for QF1

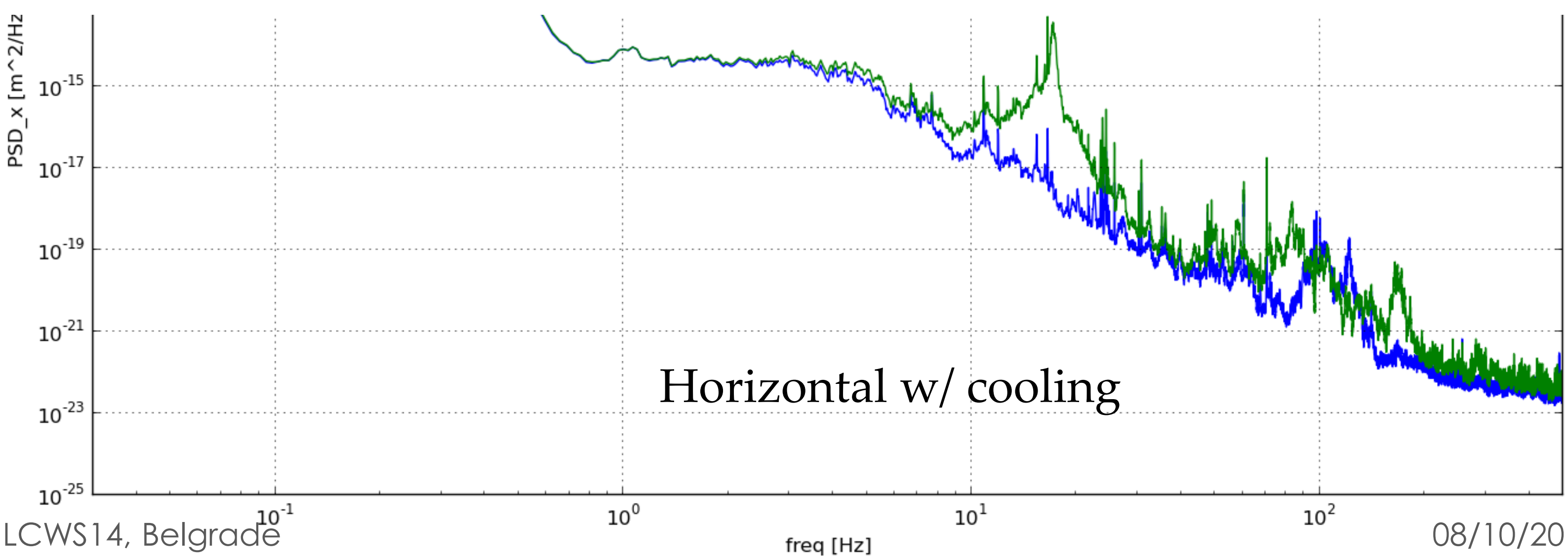


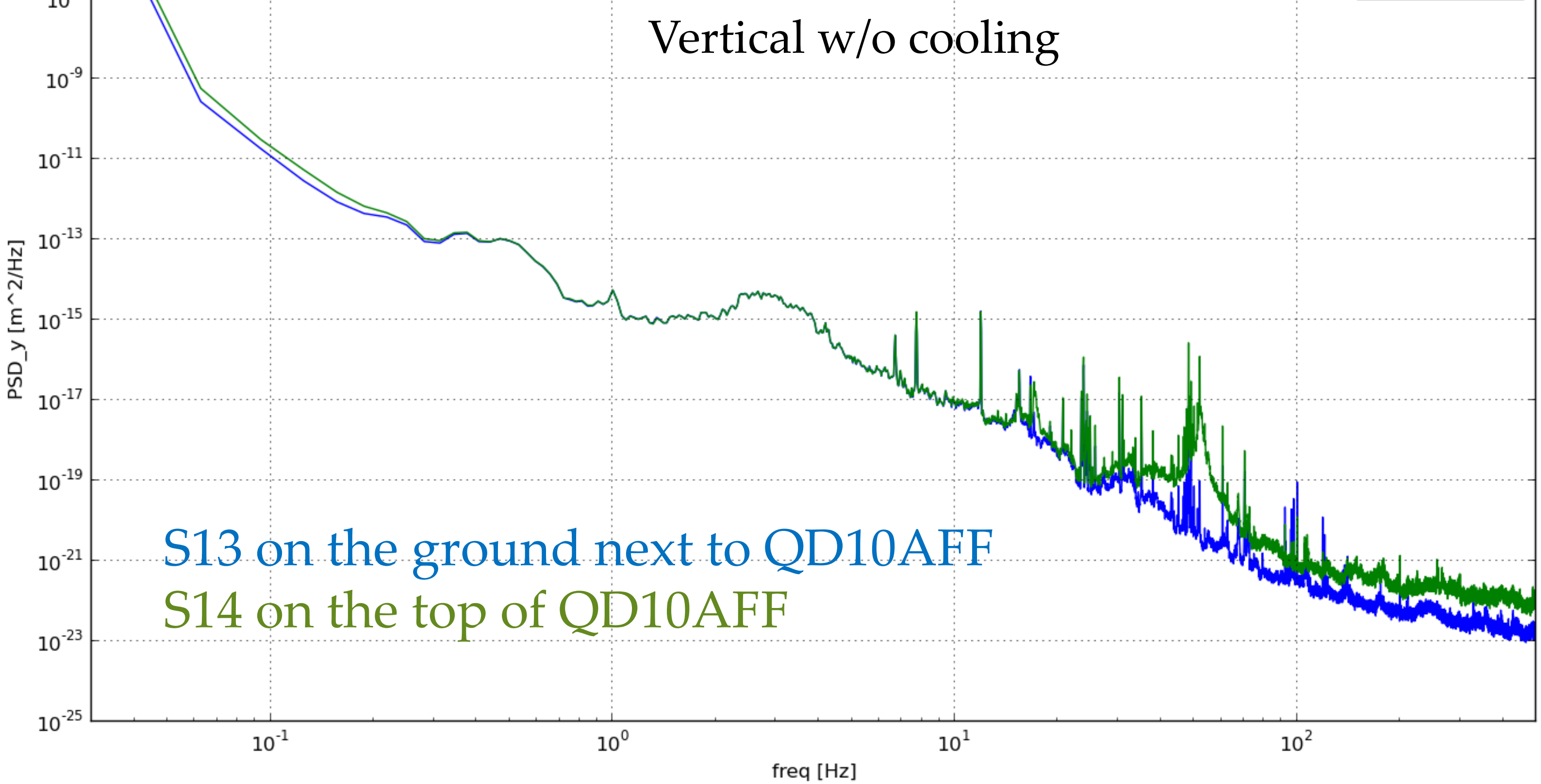
Horizontal w/o shims
8Hz
S13 on the top of QF1FF
S14 on the table of QF1FF
PSD_x [m^2/Hz]
freq [Hz]
Horizontal w/ shims
PSD_x [m^2/Hz
freq [Hz]

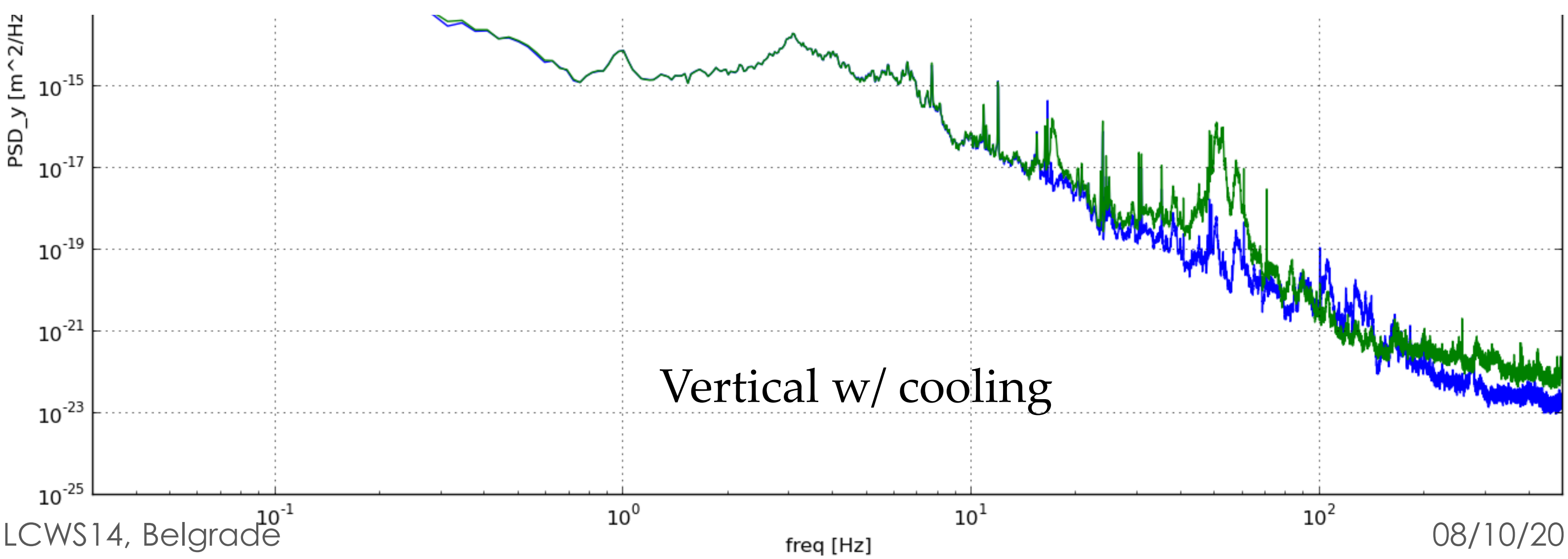

Vertical w/o shims
8Hz
S13 on the top of QF1FF
S14 on the table of QF1FF
Vertical w/ shims
PSD_y [m^2/Hz]
freq [Hz]

QD10ABFF

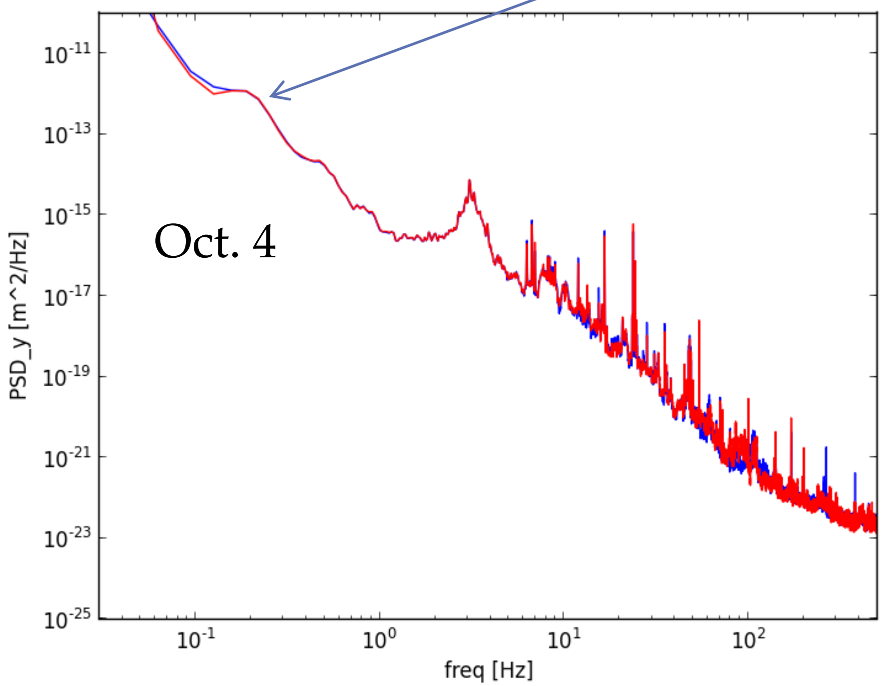

Horizontal w/o cooling
S13 on the ground next to QD10AFF
S14 on the top of QD10AFF
PSD_x [m^2/Hz]
freq [Hz]
10^{-9}
10^{-11}
10^{-13}
10^{-15}
10^{-17}
10^{-19}
10^{-21}
10^{-23}
10^{-25}
10^{-1}
10^{0}
10^{1}
10^{2}

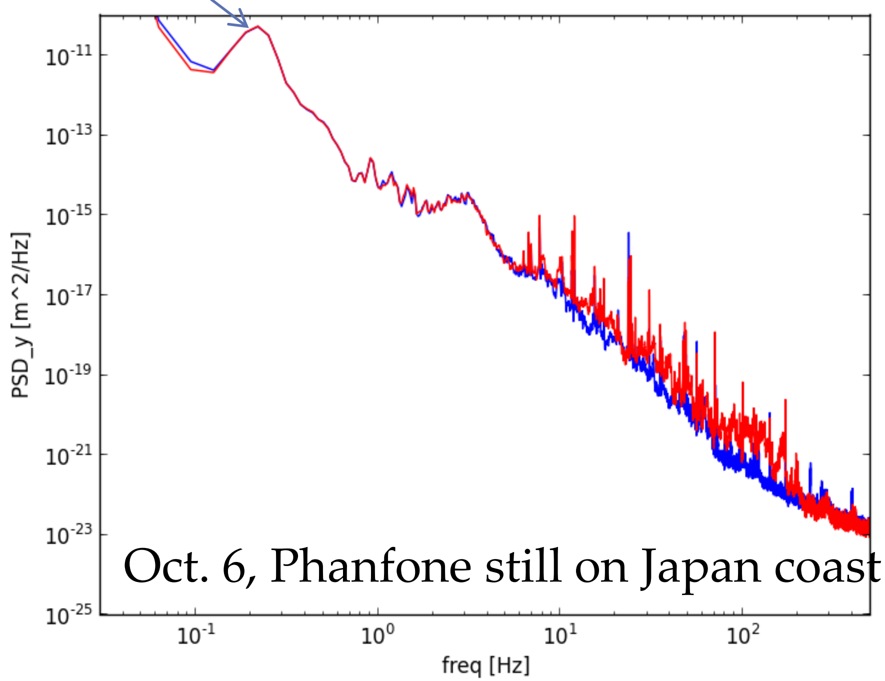

Horizontal w/ cooling
PSD_x [m^2/Hz
freq [Hz]
10^{-15}
10^{-17}
10^{-19}
10^{-21}
10^{-23}
10^{-25}
10^{-1}
10^{0}
10^{1}
10^{2}

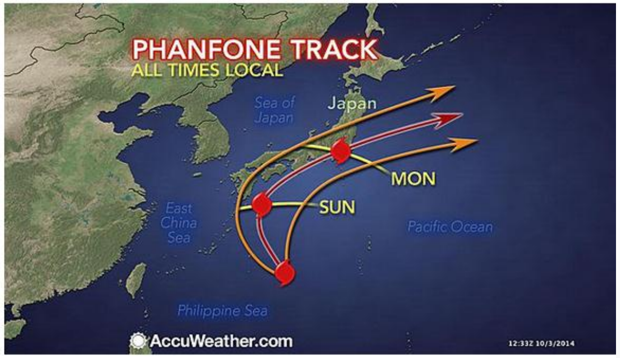

Vertical w/o cooling
S13 on the ground next to QD10AFF
S14 on the top of QD10AFF
PSD_y [m^2/Hz]
freq [Hz]


Vertical w/ cooling
PSD_y [m^2/Hz
freq [Hz]

Effect of waves called "7second hum"







PSD in vertical direction => Blue curve : near QF1FF, Red curve: near other magnet upstream

Not the same measurement, so the PSDs are different.

Thank you!

• • •